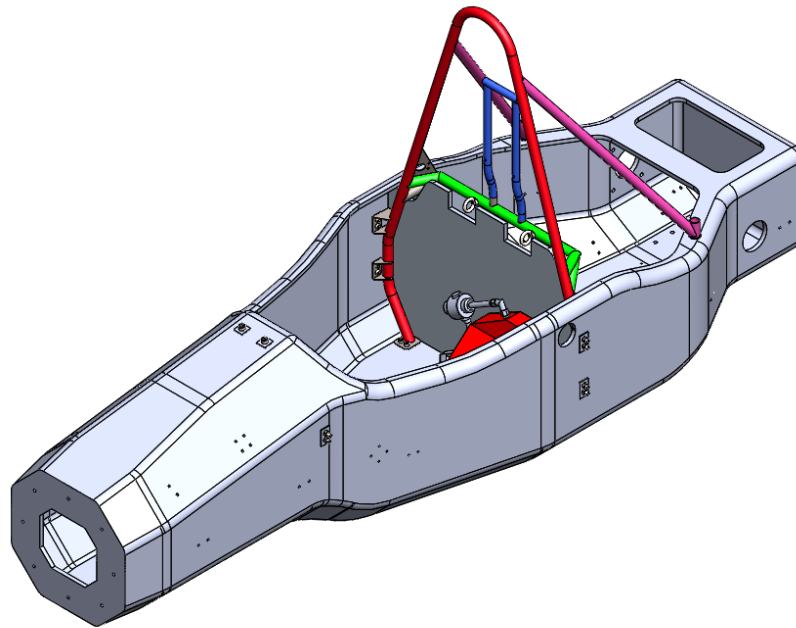


# 2023 Structural Equivalency Spreadsheet

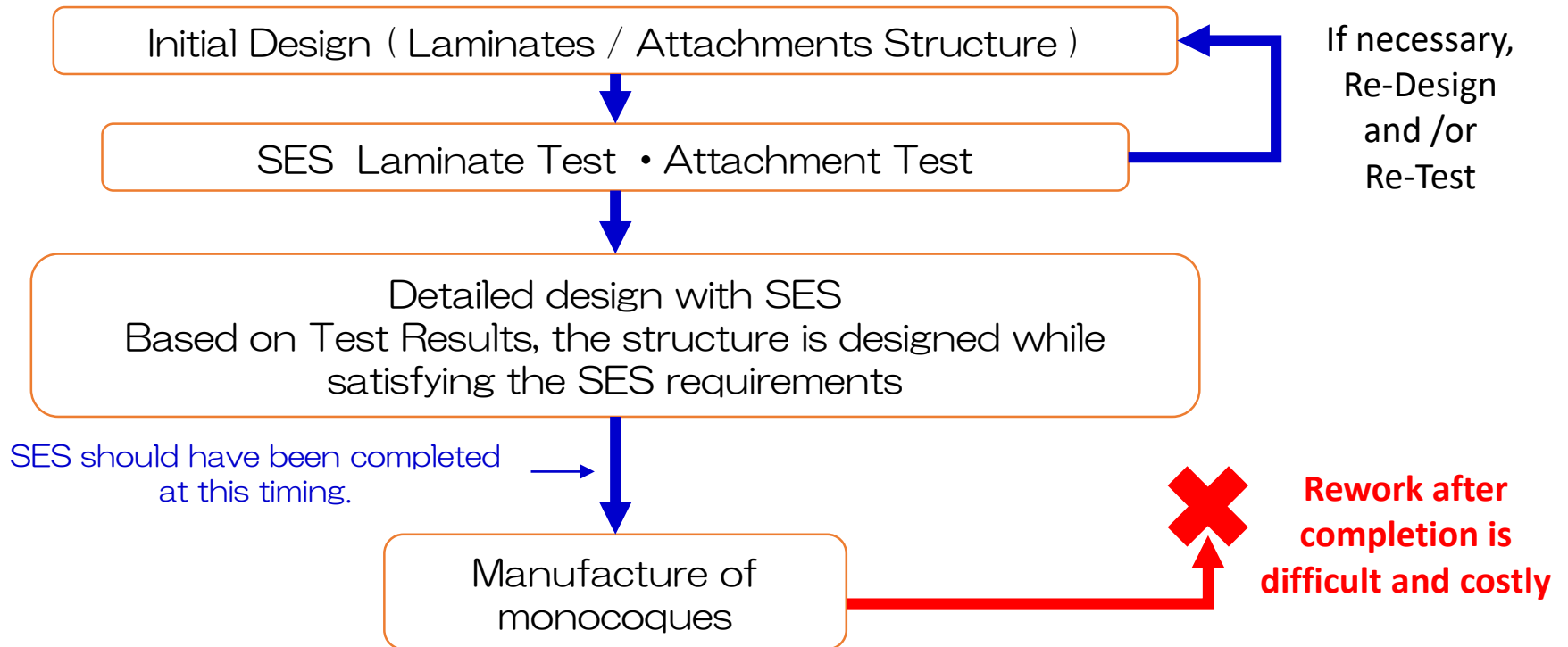
## Monocoque



# How to describe Monocoque SES

## Design flow with SES

SES requires proof tests (Laminate Test and Attachment Test) for equivalence proof  
⇒ By making good use of SES,  
you can avoid the risk of fatal rework and non-compliance with rules.



If you test after manufacturing Monocoque and find that it lacks strength or does not comply with the rules,  
**Recovery is more difficult than Steel Tube Frame, so use SES well!**

# How to describe Monocoque SES

## Basic Procedure of SES input

- ① F.3.1-5 Tube Chassis -> Basic Info & Select [Tube] or [Composite]  
Define your Composite Portion in the Structure.



- ② F.4.3 Composite  
If necessary, duplicate [F.4.3 Composite] Sheet for Different or Additional Layup  
It's strongly recommended to **be completed** before proceeding to the next step.



- ③ Test section in F.7.9-10 Attachments & in F.8 Front Protection ( & in F.10-11 EV Accumulator )  
Sometime test results affect your Chassis design



- ④ F.7 Composite Chassis, Remaining F.7.9-10 Attachments (and F.10-11 EV Accumulator )



- ⑤ Remaining F.8 Front Protection  
Front Bulkhead section requires to complete FBHS section in F.7 Composite Chassis



- ⑥ Fill in remaining BLANKs

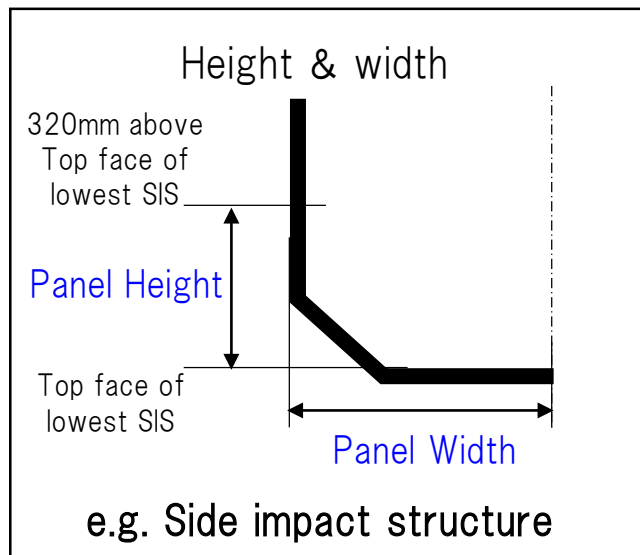
\* Of course, BLANKs may be filled when possible.

# Basic concept of Monocoque SES

## Basic Calculations of Monocoque

Monocoque Guidance Scenario 1 : Equivalent Flat Panel Calculation (F.4.4)

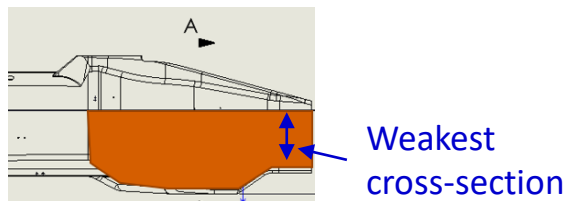
At the **weakest cross-section of structure**, the EI of the monocoque is calculated as the EI of a flat panel with the same composition as the monocoque about the neutral axis of the laminate. The curvature of the panel and geometric cross section of the monocoque must be ignored for these calculations. **It stands for safer calculation.**



Note : Comply with F.4.4 for the following calculations

- Front Bulkhead Support Structure  
**Vertical wall must have EI more than ONE Baseline steel tube.**
- Side Impact Structure  
**Vertical wall must have EI more than TWO Baseline steel tube**  
**Floor Panel must have EI more than ONE Baseline steel tube**

see (F.7.3.2, F.7.5.3, F.7.5.4)



**Note : They should be calculated about the weakest cross-section of structure.**

# **2023 Structural Equivalency Spreadsheet**

## **Monocoque**

### **F.4.3 Composite**

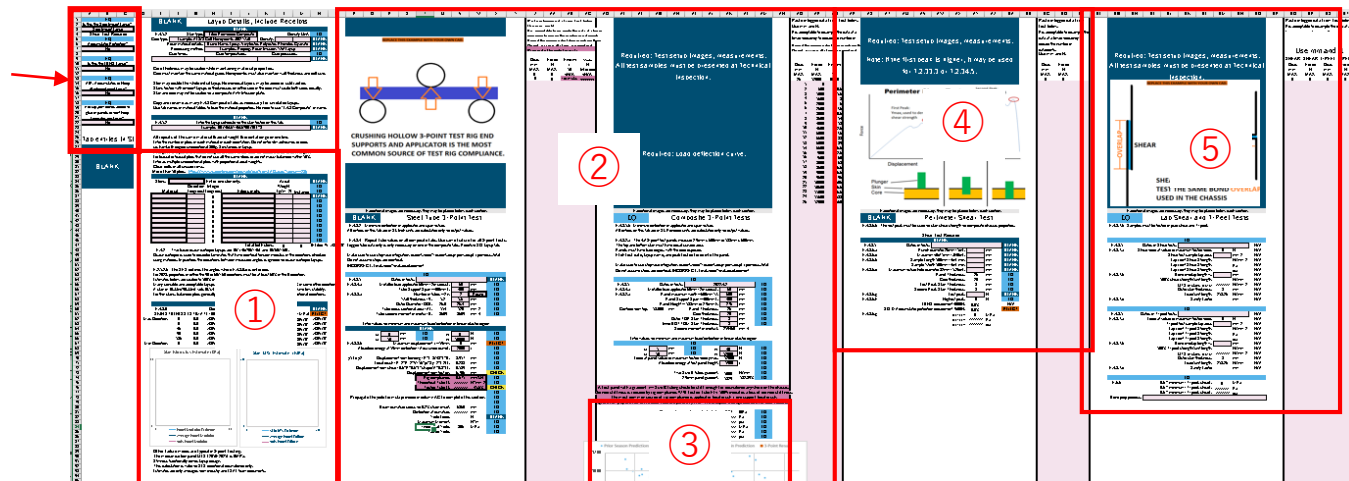
# Monocoque Laminate Test

## Derivation of Key Elements for Proof of Equivalence

If it is a tube frame, it is possible to use common values for physical properties such as Young's modulus and yield strength, but for Monocoque, physical properties vary greatly depending on how it is made, so it is essential to derive physical properties through actual tests.

Diversion of test results from different years is prohibited ( F.4.3.1b ).

Select usage  
of Laminate



- ① Quasi-Isotropic F.4.2, F4.3.6
- ② (only for SIS/Accumulator Side Protection) Equivalency Test ( F.4.3.1-F.4.3.4 )
- ③ Derived physical property value for F.7
- ④ (only for SIS/FBHS/Accumulator Protection) Shear strength F.4.3.5
- ⑤ (if using adhesion) Shear & Peel strength of adhesion F.4.3.7

# Monocoque Laminate Test

2023 revised

## ① Quasi-Isotropy

Securing quasi-isotropy has been tightened since 2023 SES.

Skin thickness in F.7 Composite Chassis has been changed from conventional actual thickness specification to scaling option

Layup Used:	SIS F.4.3 Composite	EQ
	Monocoque	EQ
Core thickness:	12 mm	EQ
Scaling option, layup repeats:	1	EQ
Outer skin thickness:	3 mm	EQ
Scaling option, layup repeats:	1	EQ
Inner skin thickness:	3 mm	EQ
Panel thickness:	18 mm	EQ

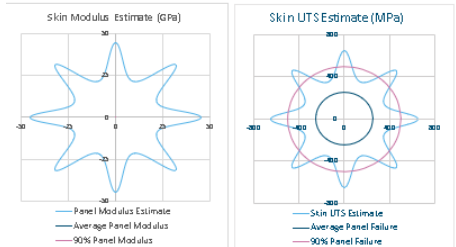
Must be an integral multiple of the Layup Schedule for Laminate Test  
(Because thickness change of 1 ply unit may lose quasi-isotropy)

Material	Direction (degree)	Integer	Fibers in ply (g/m <sup>2</sup> )	Instance
Carbon	0	0	Biaxial - Perpendicular Balance	200 2
Carbon	45	45	Biaxial - Perpendicular Balance	200 2
Carbon	90	90	Biaxial - Perpendicular Balance	200 2
Carbon	-45	-45	Biaxial - Perpendicular Balance	200 2

F.4.2 Two basic quasi-isotropic layups are [0/+45/90/-45] and [0/90/-90]. Quasi-isotropic is used to describe laminates that have identical flexion modulus in the directions checked using matrices. In practice, the reductions between measured angles is ignored for quasi-isotropic layups.

F.2.1.2.b The SES defines the angles where FA3.6.c is enforced. For 2023, properties in either the 90 or 0/90 directions must be at least 50% of the 0 direction. Estimates below are scaled to 100% in the 0 direction. Extrapolated values are discouraged. Many sensible and acceptable layups since 2019 returned a CHECK readout (below 50% in some other direction). A skin of [0\_0/+90\_0] fails G.1.4 Good Engineering Practice. Unlabeled on their own have low stability. For thin skins, balanced ones generally enclose uni and increase anisotropies in non-unidirectional directions.

Direction	SKIN STIFFNESS ESTIMATE (GPa)	SKIN STRENGTH ESTIMATE (MPa)
0	48.7 100%	Max: 0 852 100%
45	44.9 100%	EQ 0 852 100%
90	44.9 100%	EQ 45 852 100%
135	44.9 100%	EQ 90 852 100%
225	29.5 52%	Min: 119 981 98%



For example, if you subtract 1ply from [0/45/90/-45] evenly

Reduced stiffness in specific directions

⇒ Loss of quasi-isotropy

± 60deg or 90deg direction need 50% or more at 0deg (See Comment in SES)

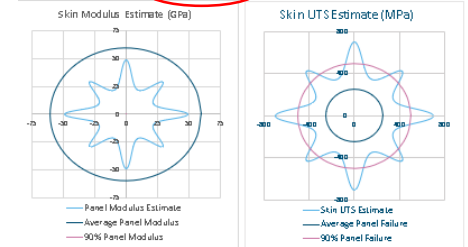
All thickness differences that are not integral multiples should be treated as Different Layups and their physical properties should be obtained using the Laminate Test.

Material	Direction (degree)	Integer	Fibers in ply (g/m <sup>2</sup> )	Instance
Carbon	0	0	Biaxial - Perpendicular Balance	200 2
Carbon	45	45	Biaxial - Perpendicular Balance	200 2
Carbon	90	90	Biaxial - Perpendicular Balance	200 2
Carbon	-45	-45	Biaxial - Perpendicular Balance	200 2

F.4.2 Two basic quasi-isotropic layups are [0/+45/90/-45] and [0/90/-90]. Quasi-isotropic is used to describe laminates that have identical flexion modulus in the directions checked using matrices. In practice, the reductions between measured angles is ignored for quasi-isotropic layups.

F.2.1.2.b The SES defines the angles where FA3.6.c is enforced. For 2023, properties in either the 90 or 0/90 directions must be at least 50% of the 0 direction. Estimates below are scaled to 100% in the 0 direction. Extrapolated values are discouraged. Many sensible and acceptable layups since 2019 returned a CHECK readout (below 50% in some other direction). A skin of [0\_0/+90\_0] fails G.1.4 Good Engineering Practice. Unlabeled on their own have low stability. For thin skins, balanced ones generally enclose uni and increase anisotropies in non-unidirectional directions.

Direction	SKIN STIFFNESS ESTIMATE (GPa)	SKIN STRENGTH ESTIMATE (MPa)
0	48.7 100%	Max: 0 792 100%
45	41.1 84%	EQ 45 891 98%
90	48.7 100%	EQ 90 792 100%
135	41.1 84%	EQ 135 891 98%
225	29.5 52%	Min: 245 980 54%



# Monocoque Laminate Test

## ②③ Equivalency Test & Laminate Test

② F.4.3.2- At the same time as proof of F.4.3.4, the specifications of ③ can be obtained

BLANK			
F.4.3.1	Dates of tests:		BLANK
F.4.3.4.a	Metallc load applicator 50mm (zin radius):		BLANK
	Tube Support Span =400mm L:		BLANK
F.4.3.3.a	Number of tubes =2 n:	Round	BLANK
	Wall thickness (t): 1.2		BLANK
	Outer Diameter (OD): 25.0		BLANK
	Tube cross sectional area (A): 114	mm <sup>2</sup>	BLANK
	Tube second moment of inertia (I): 8509	mm <sup>4</sup>	BLANK

**Test results are valid for that year only  
Historical data is invalid.**

Enter values for minimum and maximum load/deflection in linear-elastic region			
BLANK			
x <sub>1</sub>		mm	BLANK
x <sub>2</sub>		mm	BLANK
F.4.3.3.b	Maximum displacement >=19mm:	0	mm
	Absorbed energy at 19mm deflection (Area under curve):		J
y <sub>1</sub> to y <sub>2</sub>	Displacement from bending ( $P \cdot L^3 / 48 \cdot E \cdot I$ ):	#DIV/0!	mm
	Local crush ( $P^2 \cdot P^2 \cdot t / (16 \cdot \pi \cdot S_y^2 \cdot I^2)$ ):	#DIV/0!	mm
	Displacement from shear ( $0.5 \cdot P \cdot 0.5 \cdot L \cdot \text{shape} / A \cdot 0.3 \cdot E$ ):	#DIV/0!	mm
	Displacement from test rig:	#DIV/0!	mm
	Rig compliance:	#DIV/0!	mm/kN
	Theoretical EI: #VALUE!	N*mm <sup>2</sup>	
	Tested EI, uncorrected:		
Propagate the yield formula provided in column AC to complete this section.			
	Beam curvature radius for 0.2% strain offset:	0	mm
	Deflection at curvature:	0.00E+00	mm
	Yield Force:	N	
	Maximum Moment:	N*m	
	Theoretical Yield:	305	MPa
	Tested Yield:		

### F.4.3.1 Testing Requirements

- Any tested samples must be engraved with the construction date, sample name, and peak test force.
- The same set of test results must not be used for different monocoques in different years.

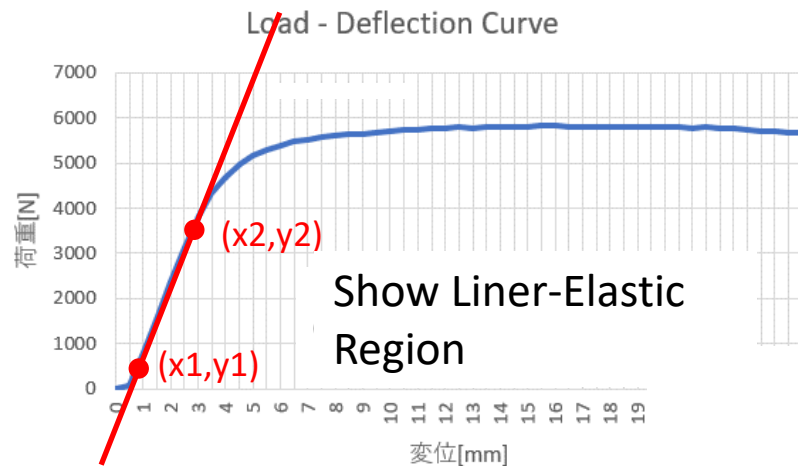
*The intent is for the test panel to use the same material batch, material age, material storage, and student layup quality as the monocoque.*

Explained on  
the next page



# Monocoque Laminate Test

Take a photo with the date in the photo



Paste in logged data from test below:  
Use mm and N.  
It is acceptable to resample the data at a lower frequency to reduce the number of datapoints.  
Repeat the energy calculation in column three.  
Do not assume all steps are identical.  
Propagate the yield formula.

Disp. mm	Force N	Energy J	Yield N
MAX	MAX	19	Intercept
25	5820	98.21	3.52E+03
0	0	Formula:	-4.73E+03
0.5	80	0.04	-4.06E+03
1	810	0.445	-4.04E+03
1.5	1610	1.25	-4.09E+03
2	2350	2.425	-4.08E+03
2.5	3130	3.99	-4.11E+03
3	3830	5.905	-4.06E+03
3.5	4330	8.07	-3.81E+03
4	4680	10.41	-3.41E+03
4.5	4980	12.9	-2.96E+03
5	5160	15.48	-2.39E+03
5.5	5300	18.13	-1.78E+03
6	5400	20.83	-1.13E+03
6.5	5470	23.565	-4.46E+02
7	5520	26.325	2.54E+02
7.5	5570	29.11	9.54E+02

Yield column applies this formula to all rows

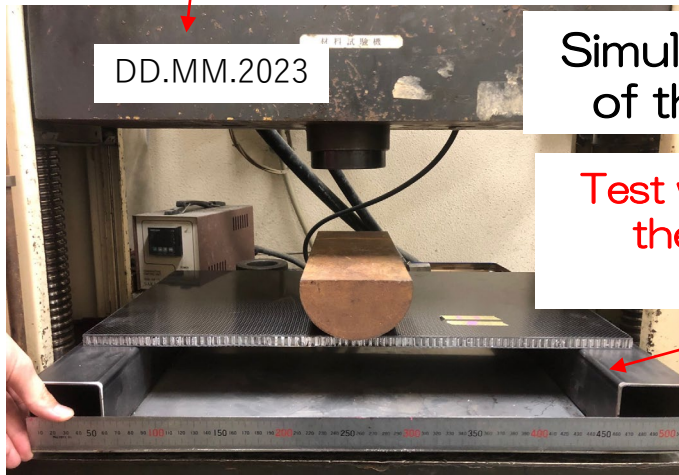
The Energy column is Integrated value (described later)

16.5	5810	80.815	1.42E+04
17	5800	83.715	1.50E+04
17.5	5810	86.62	1.57E+04
18	5800	89.52	1.65E+04
18.5	5790	92.415	1.72E+04
19	5790	95.31	1.80E+04
19.5	5800	98.21	1.87E+04
20	5800	101.11	1.95E+04
20.5	5800	104.01	2.02E+04

Raw Data must be input directly

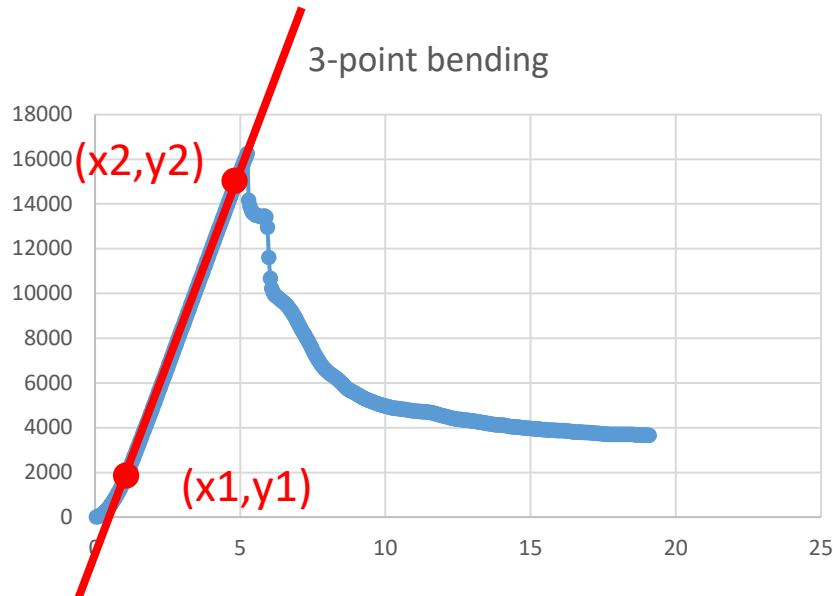
# Monocoque Laminate Test

Take a photo with the date in the photo



Simultaneous Shooting  
of the Rig and Scale

Test with the same Rig as  
the Steel Tube Test!  
(F.4.3.4)



Paate in logged data from test below:  
Use mm and N.

It is acceptable to resample the data at  
a lower frequency to reduce the number

Repeat the energy calculation in column t  
Do not assume all steps are identical.

Disp. mm	Force N	Energy J
MAX	MAX	19
21.5	14510	125.71
0	8	
0.5	203	0.1015
1	454	0.3285
1.5	705	0.687
2	980	1.171
2.5	1223	1.7825
3	1506	2.5355
3.5	1781	3.426
4	2040	4.446
4.5	2323	5.6075
5	2638	6.9265
5.5	2977	8.415
6	3340	10.085
6.5	3743	11.9565
7	4106	14.0095
7.5	4445	16.232

The Energy  
column is  
Integrated value  
(described later)

Raw Data must  
be input directly

16.5	11187	88.2535
17	11566	94.0365
17.5	11921	99.997
18	12300	106.147
18.5	12655	112.475
19	13058	119.004
19.5	13405	125.706
20	13768	132.59
21	14155	139.668
21	14510	146.923
21.5	0	146.923

# Monocoque Laminate Test

## Precautions when entering test data

Some items reflect the experimental RAW data attached, so the values of "Displacement", "Force", and "Energy" must all be entered directly.

Enter values for minimum and maximum load/deflection in linear-elastic region

#1	0.5	mm	EQ	y1	203	N	EQ
#2	20	mm	EQ	y2	13768	N	EQ
Force at panel failure or maximum tested force ymax:				14510	N		EQ
Absorbed energy at test panel height:					J		N/A
Two Size B tubes gradient:				1500	N/mm		EQ
275mm panel gradient:				696	46.38%		EQ

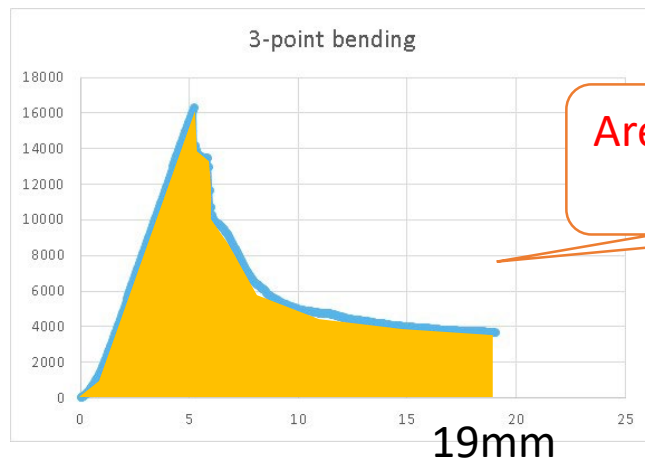
Paste in logged data from test below:  
Use mm and N.  
It is acceptable to resample the data at a lower frequency to reduce the number  
Repeat the energy calculation in column t  
Do not assume all steps are identical.

Disp. mm	Force N	Energy J
MAX	MAX	19
21.5	14510	125.71
0	8	
0.5	203	0.1015
1	454	0.3285
1.5	705	0.681
2	980	1.171
2.5	1223	1.7825
3	1506	2.5355
3.5	1781	3.426

The Energy column should be entered as the cumulative Energy as shown in the formula below.

$$\text{Energy}(n) = \text{Energy}(n-1) + (\text{Disp}(n) - \text{Disp}(n-1)) * \text{Force}(n) / 1000$$

Correctly understand the meaning of Absorbed Energy!



Area of Deflection Curve  
= Absorbed Energy

16	10792	82.66
16.5	11187	88.2535
17	11566	94.0365
17.5	11921	99.997
18	12300	106.147
18.5	12655	112.475
19	13058	119.004
19.5	13405	125.706
20	13768	132.59
21	14155	139.668
21	14510	146.923
21.5	0	146.923

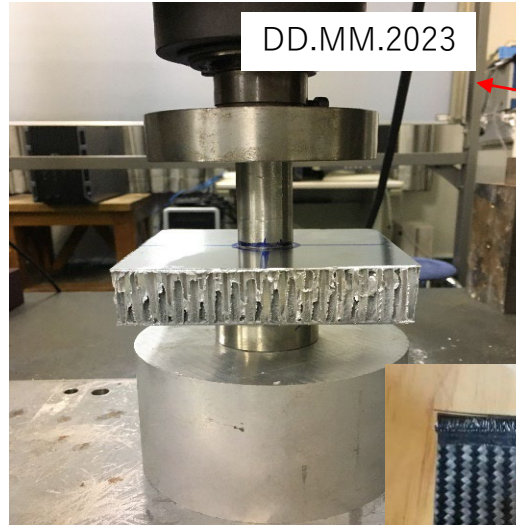
# Monocoque Laminate Test

## ④Perimetr Shear test

Required: Test setup images, measurements.

Note: If the first peak is higher, it may be used for T.2.33.3 or T.2.34.5.

Required: Load deflection curve.



Take a photo  
with the date  
in the photo

Paste in logged data from test below:

It is acceptable to resample the data at a lower frequency to reduce the number of datapoints.

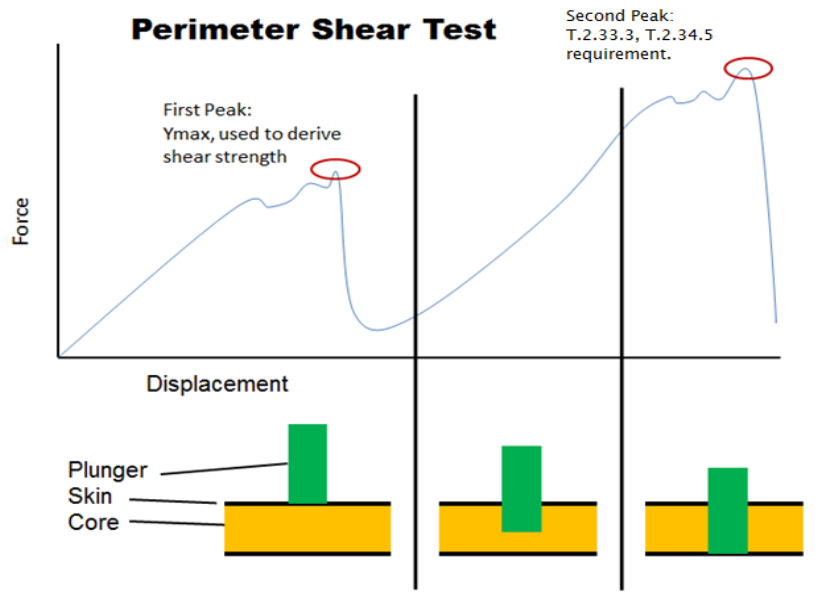
Use mm and N.

Disp. mm	Force N
MAX	MAX
0	0

Raw Data must  
be input directly

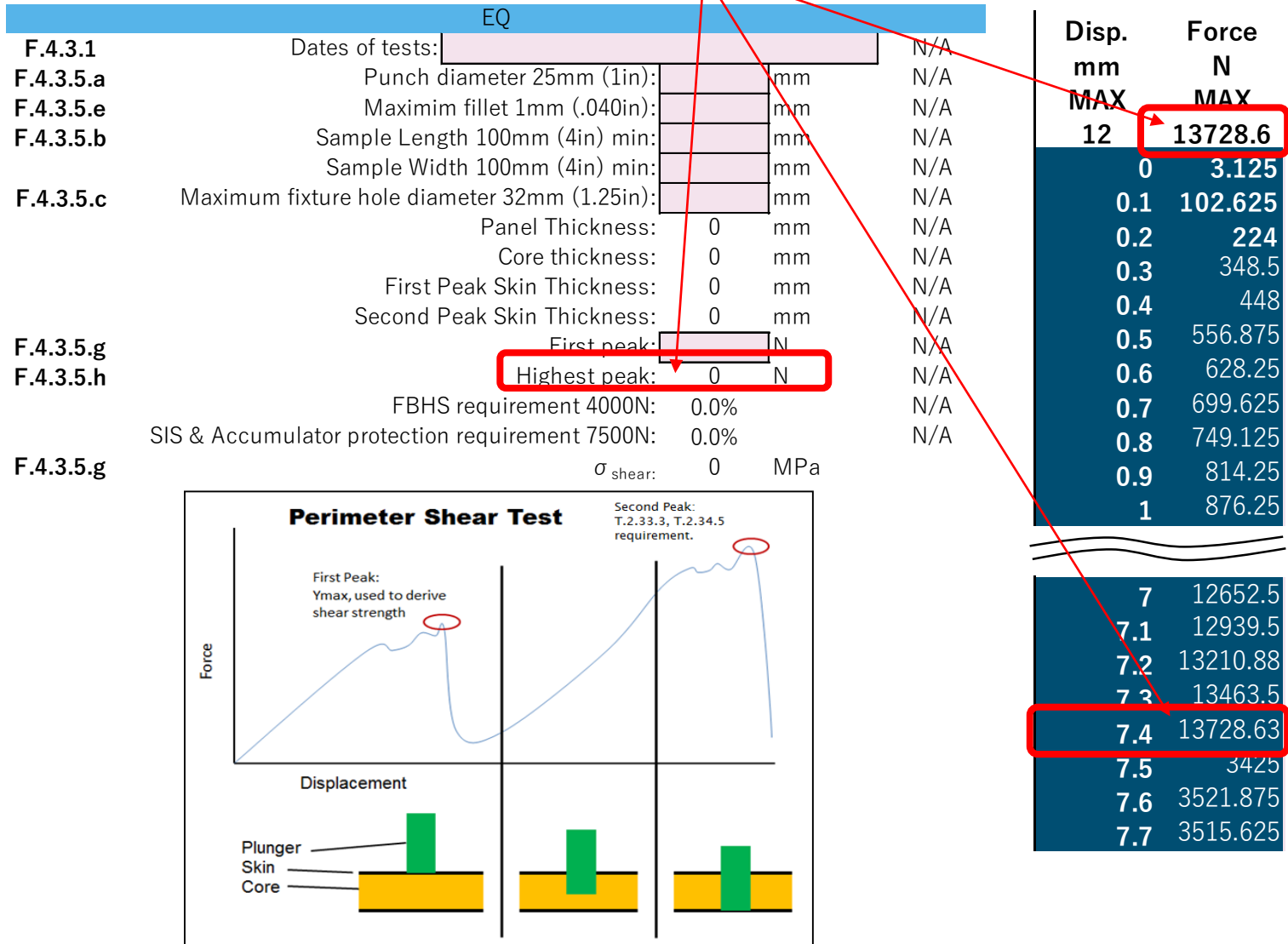


### Perimeter Shear Test



# Monocoque Laminate Test

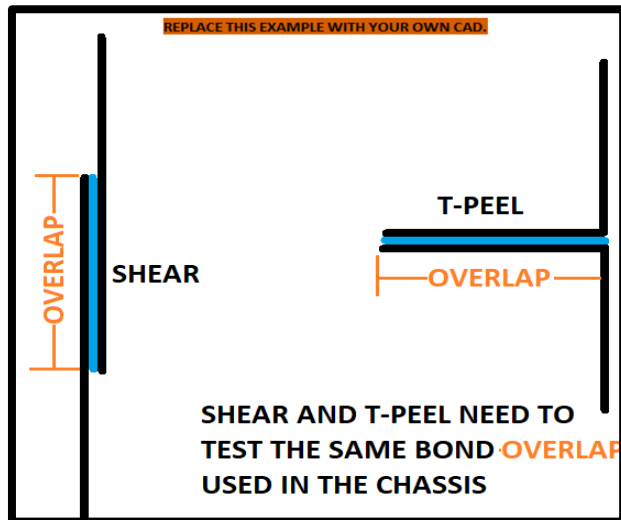
Some items reflect the experimental RAW data attached, so the values of “Displacement “ and “Force “ must all be entered directly.



# Monocoque Laminate Test

## ⑤ Lap-Joint Test

Enter test results for both Shear Test and T-Peel Test



Enter pretreatment for adhesion in  
「Bond prep Process」  
(e.g. polishing / degreasing)

EQ				
F.4.3.1	Dates of Shear tests:			N/A
F.4.3.7.a	Shear force at failure or maximum tested force:	0	N	N/A
	Shear test sample lap area:		mm <sup>2</sup>	N/A
	Lap Joint Shear Strength:		N/mm <sup>2</sup>	N/A
	Lap Joint Shear Strength:		Pa	N/A
	Lap Joint Shear Strength:		psi	N/A
F.4.3.7.b	Bond overlap length w:		mm	N/A
	100% shear strength/unit length:		N/mm	N/A
	UTS of skins $\sigma_{UTS}$ :	#DIV/0!	N/mm <sup>2</sup>	N/A
	Outer skin thickness:	0	mm	N/A
	Load/unit length:	#DIV/0!	N/mm	N/A
F.4.3.7.d	Safety Factor		mm	N/A

EQ				
F.4.3.1	Dates of T-peel tests:			N/A
F.4.3.7.a	Force at failure or maximum tested force:	0	N	N/A
	T-peel test sample lap area:		mm <sup>2</sup>	N/A
	Lap Joint T-peel Strength:		N/mm <sup>2</sup>	N/A
	Lap Joint T-peel Strength:		Pa	N/A
	Lap Joint T-peel Strength:		psi	N/A
F.4.3.7.b	Bond overlap length w:		mm	N/A
	100% T-peel strength/unit length:		N/mm	N/A
	UTS of skins $\sigma_{UTS}$ :	#DIV/0!	N/mm <sup>2</sup>	N/A
	Outer skin thickness:	0	mm	N/A
	Load/unit length:	#DIV/0!	N/mm	N/A
F.4.3.7.d	Safety Factor		mm	N/A

EQ				
F.5.5	0.5 * minimum (T-peel, shear):	0	MPa	
	0.5 * minimum (T-peel, shear):	#####	Pa	
	0.5 * minimum (T-peel, shear):	#####	psi	

Bond prep process:

# Monocoque Laminate Test

## A. Different or Additional Layup

If there are multiple types of layups, duplicate the SES F.4.3 Composite sheet and enter the test results each time

How to refer

F.7 Composite Chassis

Enter each sheet name in A4:B20 and select from the pull-down menu of each [Layup Used:]

Note: Forces are given in Pa, not Mpa or Gpa.				Tested	Tested	Tested
Material	E (Pa)	S_Ultimate (Pa)	Shear (Pa)	Core	Outer	Inner
F.3.4.2 Steel	2.00E+11	3.65E+08	2.11E+08	mm	mm	mm
SIS F.4.3 Composite	8.07E+09	7.11E+07	0.00E+00	20	3	3
MHBS F.4.3 Composite	3.34E+10	1.78E+08	3.13E+07	15	2	2

REJECT			
F.7.4	Front Bulkhead Support Construction:	Composite	EQ
	Size C Steel Tubes Replaced On One Side:	3	EQ
	Layup Used:	MHBS F.4.3	EQ
		SIS F.4.3 Composite	EQ
		MHBS F.4.3 Composite	EQ
	Core thickness:		EQ
	Outer skin thickness:		EQ
	Inner skin thickness:		EQ
	Panel thickness:		EQ
	Side View Height (Minus holes and single skins):		EQ

## F.7.9-10 Attachments / F.8 Front Protection

Enter the sheet name directly in each [Type SES Tab Name Of Layup Used]

EQ			
F.7.5	Front Hoop Mounts:	Composite	EQ
	Front Hoop Mounts:	Bolted	EQ
Skin used: Both	Type SES Tab Name Of Layup Used:	MHBS F.4.3 Composite	N/A
	Front Hoop centerline length:	1000 mm	N/A
Scaling option, layup repeats: 2	Laminate thickness:	8 mm	N/A
	Skin shear area - centerline x 1 thickness:	0.008 m <sup>2</sup>	N/A
	Skin shear strength:	3.13E+07 Pa	N/A

# **2023 Structural Equivalency Spreadsheet**

## **Monocoque**

### **F.7 Composite Chassis**



# F.7 Composite Chassis

## Summary of F.7 Composite Chassis sheet

Material		Main Dimensions (mm)		Weight (kg)		Cost (£)		Notes	
Part	Material	Length	Width	Height	Weight	Cost	Notes	Notes	
1	FRP	1000	1000	1000	1000	1000	FRP	FRP	
2	FRP	1000	1000	1000	1000	1000	FRP	FRP	
3	FRP	1000	1000	1000	1000	1000	FRP	FRP	
4	FRP	1000	1000	1000	1000	1000	FRP	FRP	
5	FRP	1000	1000	1000	1000	1000	FRP	FRP	
6	FRP	1000	1000	1000	1000	1000	FRP	FRP	
7	FRP	1000	1000	1000	1000	1000	FRP	FRP	
8	FRP	1000	1000	1000	1000	1000	FRP	FRP	
9	FRP	1000	1000	1000	1000	1000	FRP	FRP	
10	FRP	1000	1000	1000	1000	1000	FRP	FRP	
11	FRP	1000	1000	1000	1000	1000	FRP	FRP	
12	FRP	1000	1000	1000	1000	1000	FRP	FRP	
13	FRP	1000	1000	1000	1000	1000	FRP	FRP	
14	FRP	1000	1000	1000	1000	1000	FRP	FRP	
15	FRP	1000	1000	1000	1000	1000	FRP	FRP	
16	FRP	1000	1000	1000	1000	1000	FRP	FRP	
17	FRP	1000	1000	1000	1000	1000	FRP	FRP	
18	FRP	1000	1000	1000	1000	1000	FRP	FRP	
19	FRP	1000	1000	1000	1000	1000	FRP	FRP	
20	FRP	1000	1000	1000	1000	1000	FRP	FRP	
21	FRP	1000	1000	1000	1000	1000	FRP	FRP	
22	FRP	1000	1000	1000	1000	1000	FRP	FRP	
23	FRP	1000	1000	1000	1000	1000	FRP	FRP	
24	FRP	1000	1000	1000	1000	1000	FRP	FRP	
25	FRP	1000	1000	1000	1000	1000	FRP	FRP	
26	FRP	1000	1000	1000	1000	1000	FRP	FRP	
27	FRP	1000	1000	1000	1000	1000	FRP	FRP	
28	FRP	1000	1000	1000	1000	1000	FRP	FRP	
29	FRP	1000	1000	1000	1000	1000	FRP	FRP	
30	FRP	1000	1000	1000	1000	1000	FRP	FRP	
31	FRP	1000	1000	1000	1000	1000	FRP	FRP	
32	FRP	1000	1000	1000	1000	1000	FRP	FRP	
33	FRP	1000	1000	1000	1000	1000	FRP	FRP	
34	FRP	1000	1000	1000	1000	1000	FRP	FRP	
35	FRP	1000	1000	1000	1000	1000	FRP	FRP	
36	FRP	1000	1000	1000	1000	1000	FRP	FRP	
37	FRP	1000	1000	1000	1000	1000	FRP	FRP	
38	FRP	1000	1000	1000	1000	1000	FRP	FRP	
39	FRP	1000	1000	1000	1000	1000	FRP	FRP	
40	FRP	1000	1000	1000	1000	1000	FRP	FRP	
41	FRP	1000	1000	1000	1000	1000	FRP	FRP	
42	FRP	1000	1000	1000	1000	1000	FRP	FRP	
43	FRP	1000	1000	1000	1000	1000	FRP	FRP	
44	FRP	1000	1000	1000	1000	1000	FRP	FRP	
45	FRP	1000	1000	1000	1000	1000	FRP	FRP	
46	FRP	1000	1000	1000	1000	1000	FRP	FRP	
47	FRP	1000	1000	1000	1000	1000	FRP	FRP	
48	FRP	1000	1000	1000	1000	1000	FRP	FRP	
49	FRP	1000	1000	1000	1000	1000	FRP	FRP	
50	FRP	1000	1000	1000	1000	1000	FRP	FRP	
51	FRP	1000	1000	1000	1000	1000	FRP	FRP	
52	FRP	1000	1000	1000	1000	1000	FRP	FRP	
53	FRP	1000	1000	1000	1000	1000	FRP	FRP	
54	FRP	1000	1000	1000	1000	1000	FRP	FRP	
55	FRP	1000	1000	1000	1000	1000	FRP	FRP	
56	FRP	1000	1000	1000	1000	1000	FRP	FRP	
57	FRP	1000	1000	1000	1000	1000	FRP	FRP	
58	FRP	1000	1000	1000	1000	1000	FRP	FRP	
59	FRP	1000	1000	1000	1000	1000	FRP	FRP	
60	FRP	1000	1000	1000	1000	1000	FRP	FRP	
61	FRP	1000	1000	1000	1000	1000	FRP	FRP	
62	FRP	1000	1000	1000	1000	1000	FRP	FRP	
63	FRP	1000	1000	1000	1000	1000	FRP	FRP	
64	FRP	1000	1000	1000	1000	1000	FRP	FRP	
65	FRP	1000	1000	1000	1000	1000	FRP	FRP	
66	FRP	1000	1000	1000	1000	1000	FRP	FRP	
67	FRP	1000	1000	1000	1000	1000	FRP	FRP	
68	FRP	1000	1000	1000	1000	1000	FRP	FRP	
69	FRP	1000	1000	1000	1000	1000	FRP	FRP	
70	FRP	1000	1000	1000	1000	1000	FRP	FRP	
71	FRP	1000	1000	1000	1000	1000	FRP	FRP	
72	FRP	1000	1000	1000	1000	1000	FRP	FRP	
73	FRP	1000	1000	1000	1000	1000	FRP	FRP	
74	FRP	1000	1000	1000	1000	1000	FRP	FRP	
75	FRP	1000	1000	1000	1000	1000	FRP	FRP	
76	FRP	1000	1000	1000	1000	1000	FRP	FRP	
77	FRP	1000	1000	1000	1000	1000	FRP	FRP	
78	FRP	1000	1000	1000	1000	1000	FRP	FRP	
79	FRP	1000	1000	1000	1000	1000	FRP	FRP	
80	FRP	1000	1000	1000	1000	1000	FRP	FRP	
81	FRP	1000	1000	1000	1000	1000	FRP	FRP	
82	FRP	1000	1000	1000	1000	1000	FRP	FRP	
83	FRP	1000	1000	1000	1000	1000	FRP	FRP	
84	FRP	1000	1000	1000	1000	1000	FRP	FRP	
85	FRP	1000	1000	1000	1000	1000	FRP	FRP	
86	FRP	1000	1000	1000	1000	1000	FRP	FRP	
87	FRP	1000	1000	1000	1000	1000	FRP	FRP	
88	FRP	1000	1000	1000	1000	1000	FRP	FRP	
89	FRP	1000	1000	1000	1000	1000	FRP	FRP	
90	FRP	1000	1000	1000	1000	1000	FRP	FRP	
91	FRP	1000	1000	1000	1000	1000	FRP	FRP	
92	FRP	1000	1000	1000	1000	1000	FRP	FRP	
93	FRP	1000	1000	1000	1000	1000	FRP	FRP	
94	FRP	1000	1000	1000	1000	1000	FRP	FRP	
95	FRP	1000	1000	1000	1000	1000	FRP	FRP	
96	FRP	1000	1000	1000	1000	1000	FRP	FRP	
97	FRP	1000	1000	1000	1000	1000	FRP	FRP	
98	FRP	1000	1000	1000	1000	1000	FRP	FRP	
99	FRP	1000	1000	1000	1000	1000	FRP	FRP	
100	FRP	1000	1000	1000	1000	1000	FRP	FRP	

Legend Color

1

2

Front 3/4 3D CAD

Rear 3/4 3D CAD

Side view

Top view

FHBS CAD dimensions

FHB CAD dimensions

SIS CAD dimensions

MHBS CAD dimensions

ASP CAD dimensions

TSP CAD dimensions

RI CAD dimensions

Bottom view

EV1

EV2

EV3

3\*

4\*

5\*

Fill in BLANKs where Selected 'Monocoque' in F.3.1-5 Tube Chassis

① Physical property value for each Layout (Enter F.4.3 Composite sheet name)

② Describe the color coding and meaning of colors for each layout of each figure

③ FHBS Equivalency (③\* stands for Steering Protection)

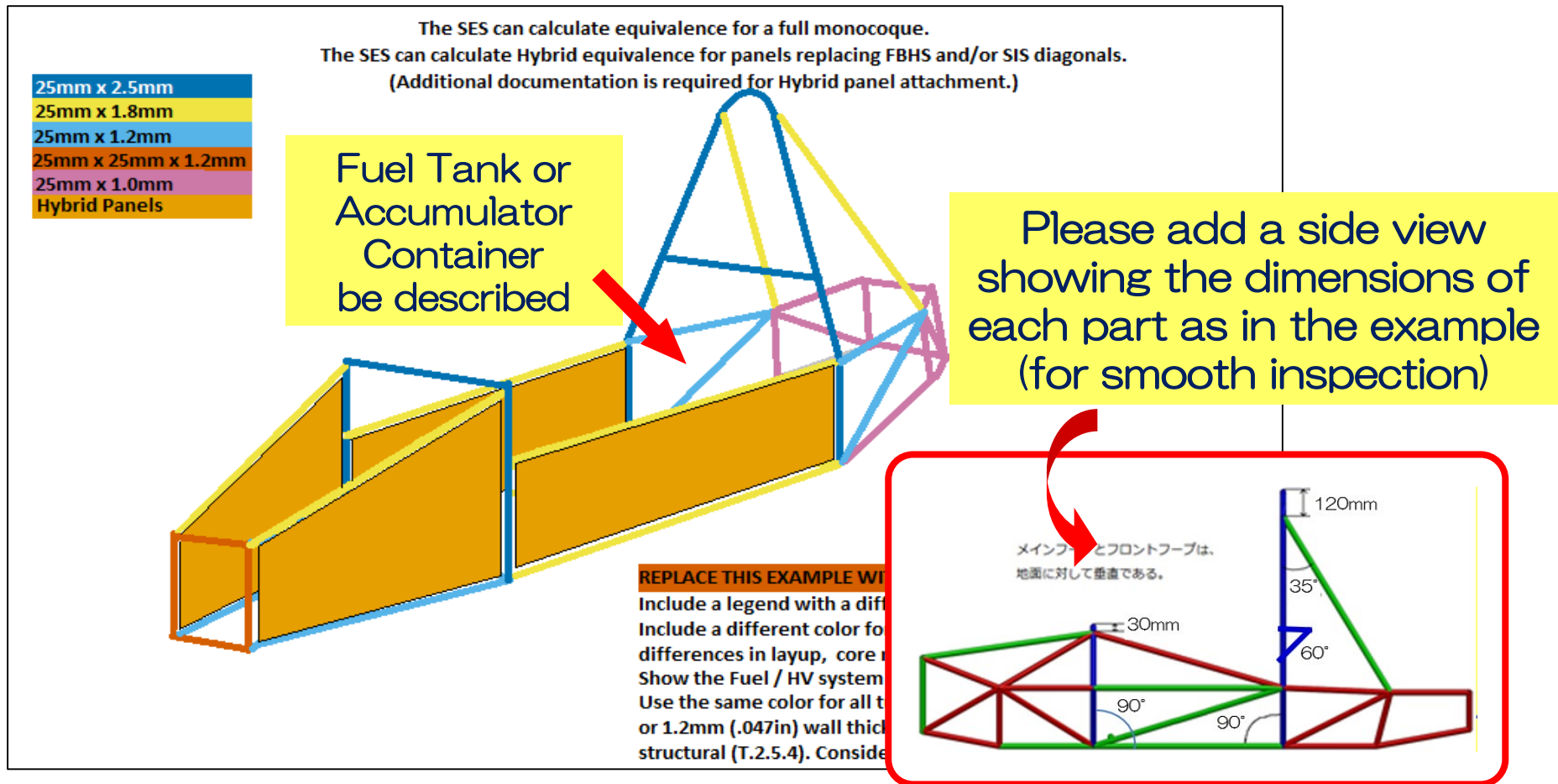
④ FHB Equivalency (④\* stands for Rearward FHB if necessary)

⑤ SIS Equivalency (⑤\* stands for the case of laminated floor)

EV①-③ EV Protection Equivalency.

# F.7 Composite Chassis

## Entry example of $\frac{3}{4}$ CAD



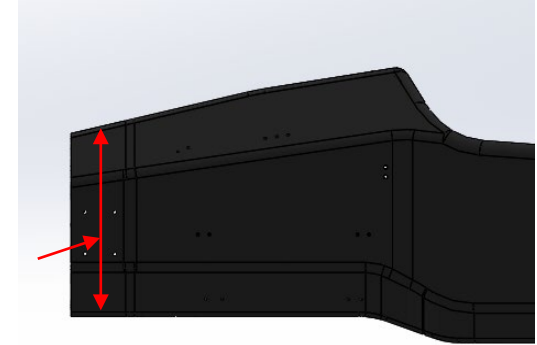
1. Show the requested CAD
2. Fuel Tank or Accumulator Container be described
3. It is recommended that the color coding of the pipe is the same as the sample.
4. All pipes with an outer diameter of <25 mm or a wall thickness of <1.2 mm Should be of the same color.

# Front Bulkhead Supports (FBHS)

## (1) Flat panel calculation

⇒ Based on Side view height, it will be evaluated as equivalent to 3 Size C steel tubes.  
Indicate that the entered Panel Height is the weakest dimension  
(If there is an opening, subtract its dimension)

Side View Height at  
Weakest points

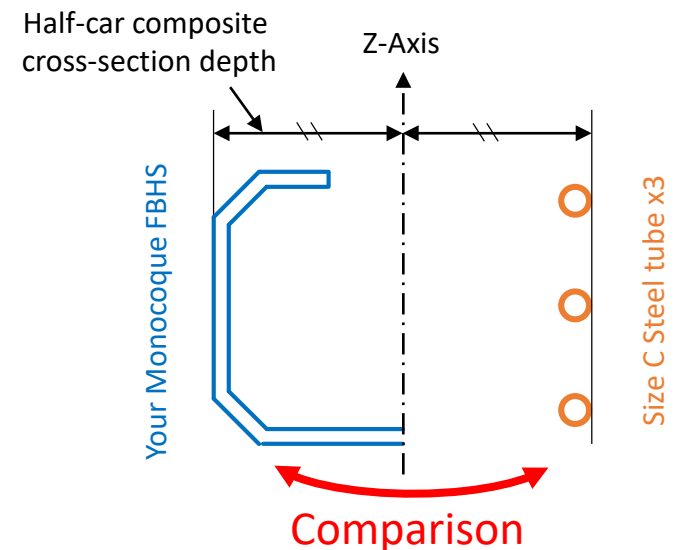


## (2) If equivalence is less than 100% in (1), use OPTION - Half Car ~

Enter **the Cross-sectional area of skin only** to  
“Half-car composite cross sectional area”

Enter **the Outer Width from car center axis** to  
“Half-car composite cross-section depth”  
(refer to right fig.)

Enter **the area moment of inertia Izz for only the skin**  
**around the vehicle center axis** (Z axis) to “Half-car  
composite second moment about car centerline”

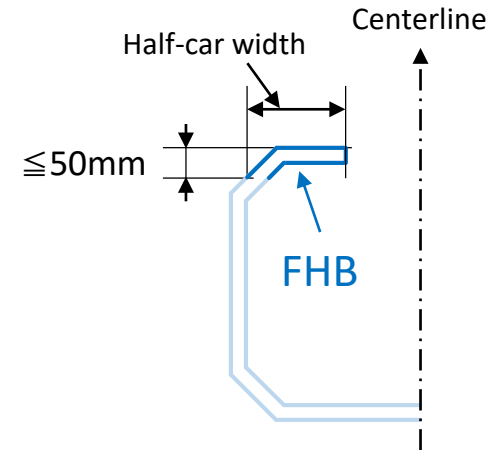


# Front Hoop Braces (FHB)

## (1) Flat panel calculation

⇒Based Half-car width, it will be evaluated as equivalent to Size B steel tubes.

Indicate that the entered Panel Width is the weakest dimension  
(If there is an opening, subtract its dimension)



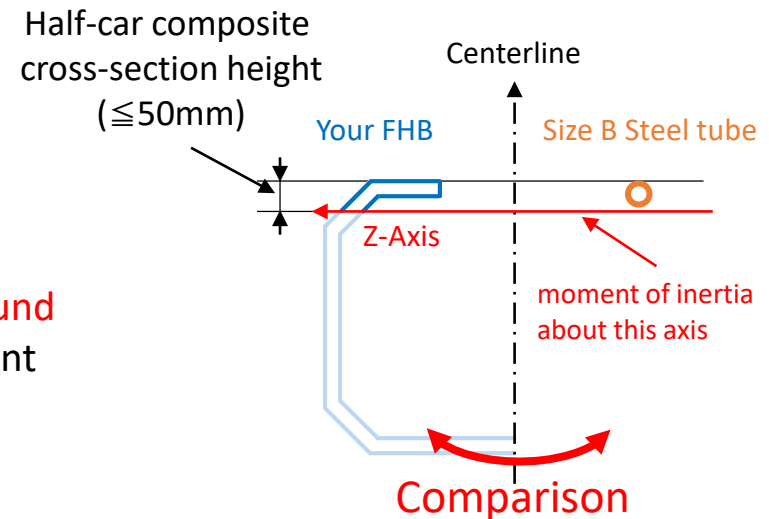
## (2) If equivalence is less than 100% in (1), use OPTION - Half Car ~

Note that there is an error in the wording of the reference axis in SES

Enter **the Cross-sectional area of skin only** to  
“Half-car composite cross sectional area”

Enter **FBH height from top ( $\leq 50\text{mm}$ )** to  
“Half-car composite cross-section height”

Enter **the area moment of inertia Izz for only the skin around Z axis in right figure** to “Half-car composite second moment about car centerline”



# Side Impact Structure (SIS)

## (1) Flat panel calculation

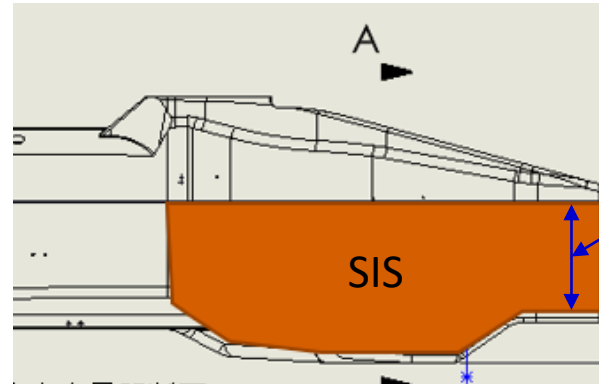
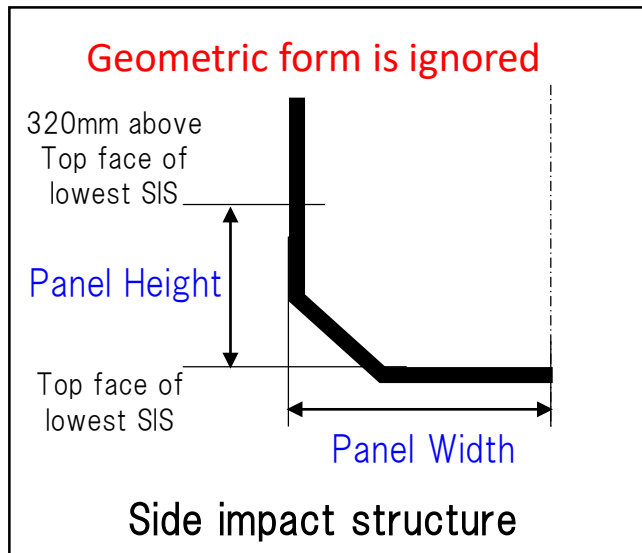
Unlike other areas of the monocoque, no allowance for geometric form is allowed.

Based on Side view Height & Half floor width, they will be evaluated as equivalent

Vertical Wall vs Size B Steel tube x2

Horizontal Wall vs Size B Steel tube x1

(If there is an opening, subtract its dimension)



Side View  
Height at  
Weakest  
points

SIS is the most important Driver Protection next to Roll Hoop among Primary Structures, so be sure to prove equivalence based on Rules!

# Main Hoop Brace Support (MHBS)

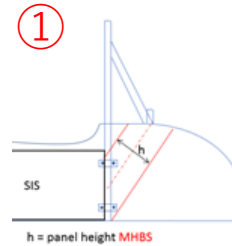
## (1) Flat panel calculation

⇒ Based on Side view height, it will be evaluated as equivalent to 2 Size C steel tubes.  
Indicate that the entered Panel Height is the weakest dimension  
(If there is an opening, subtract its dimension)

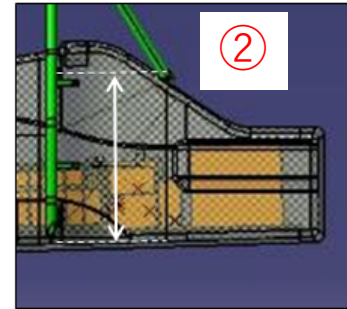
Usually use the **dimension h of ①**

Monocoque that extend below the MHB to flat floor may use Panel Height of ②

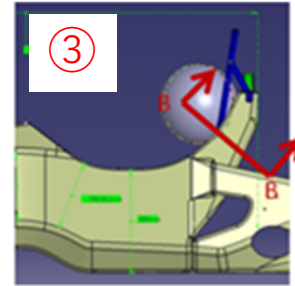
③ refer to (2) OPTION calculation



Ideally, monocoques orient their strength from midway between the MH mounts to the MHB.



Monocoques that extend below the MHB to a flat floor may use this flat panel height.



Monocoques with a limited path must use the minimum section, and are strongly encouraged to monitor laminate directional strength.

## (2) If equivalence is less than 100% in (1), use OPTION - Half Car ~

Same as FBHS

# Accumulator Side Protection

## Tractive Side Protection

## Rear Impact Protection

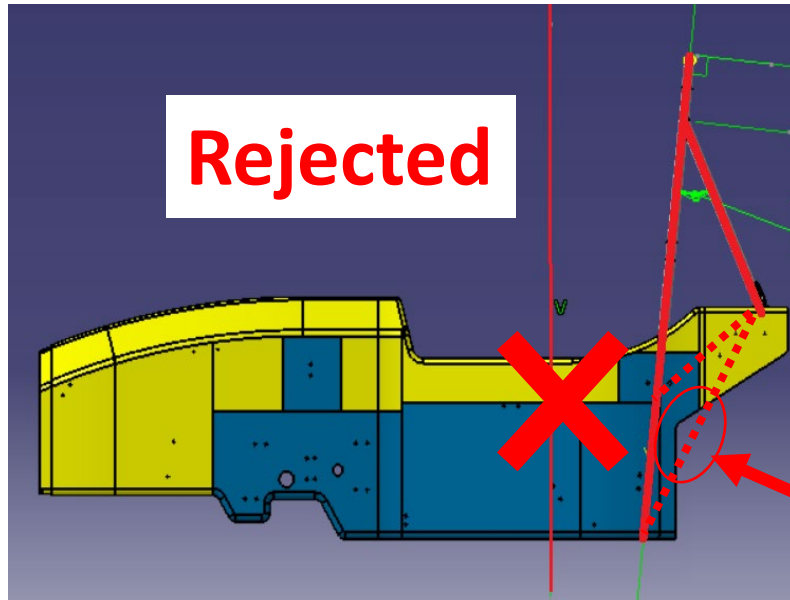
### (1) Flat panel calculation

⇒ Based on input dimension, it will be evaluated as equivalent steel tubes.  
Indicate that the entered dimension is the weakest.  
(If there is an opening, subtract its dimension)

### (2) If equivalence is less than 100% in (1), use OPTION - Half Car ~

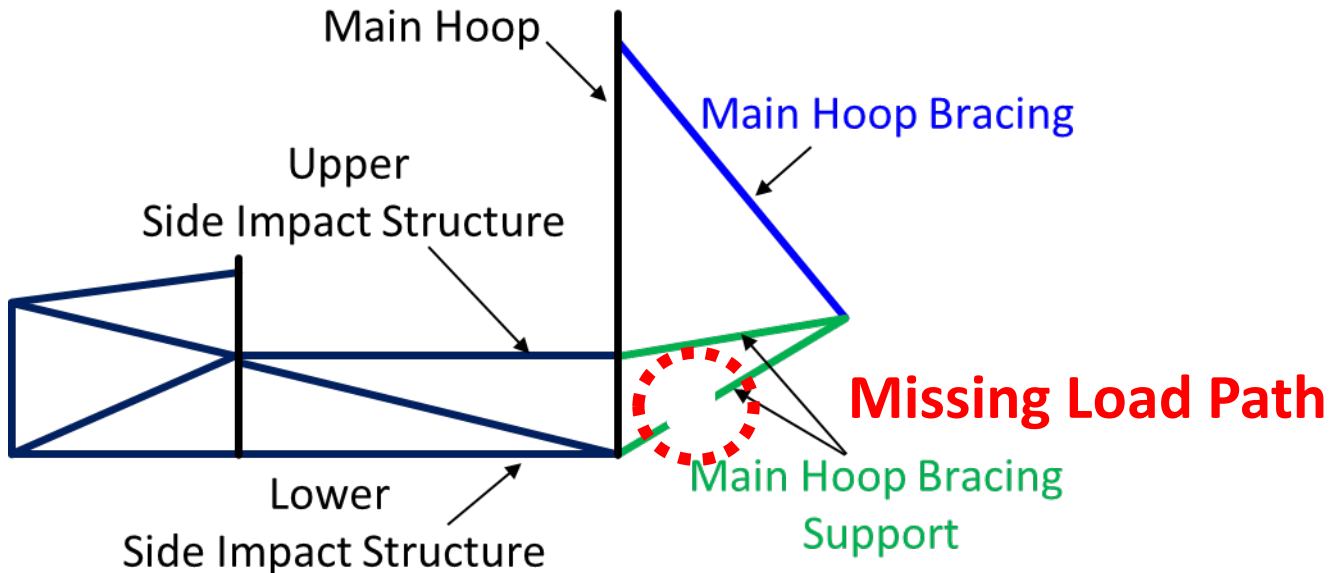
Same as FBHS

# Rejected Sample



As shown in the CAD drawing, if there is a Missing at the connection of the Lower Main Hoop Bracing Support and the Lower Side Impact Structure, it is not recognized the Load Path

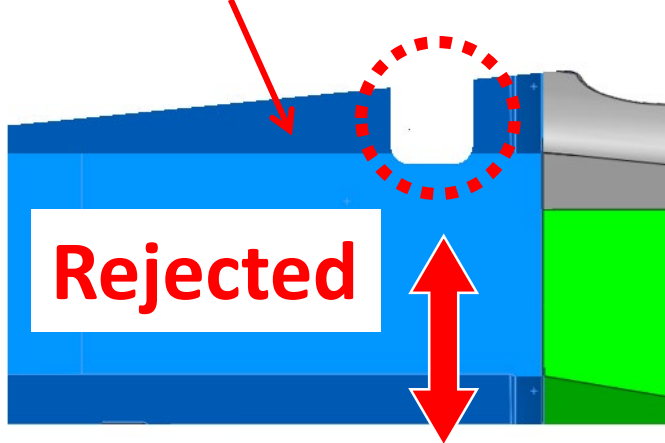
**Composite material is insufficient.**





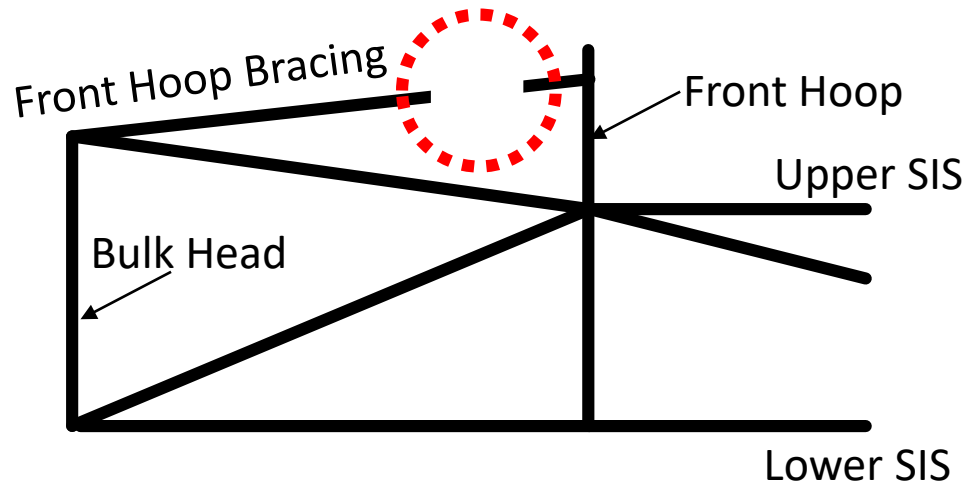
# Rejected Sample

Front Hoop Bracing  $\leq 50\text{mm}$  from the Top



As shown in the CAD drawing above, if the part corresponding to the Front Hoop Bracing of the Mnocoque structure is cut to fix the damper, it will be rejected because it is considered that the FHB has been cut.

Front Hoop Bracing is Broken...



# **2023 Structural Equivalency Spreadsheet**

## **Monocoque**

**F.10-11 Attachments**

# Attachments

[illegible]

## Actual Test section Of Harness Attachments

S.Harness Structure CAD/dimensions									
<p><i>For material &amp; design, see corresponding page for the harness listed within section.</i></p> <p><b>EO</b>      <b>Standard Harness Structure, Issues 1-5</b></p>									
<p><i>For use per page listed in table.</i></p>									
2-325A	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable - 2 Shot - 1/2" x 1/8"	Standard P	10						
	Type 212 1/2" x 3/16" x 1/8"	Standard P	10						
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
<p><i>George P. and Thomas - 1/2" x 3/16" x 1/8"</i></p>									
2-325B	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
<p><i>Shunt Cable - 1/2" x 3/16" x 1/8"</i></p>									
2-325C	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
<p><i>Shunt Cable - 1/2" x 3/16" x 1/8"</i></p>									
2-325D	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
<p><i>Shunt Cable - 1/2" x 3/16" x 1/8"</i></p>									
2-325E	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P	10					
<p><i>Shunt Cable - 1/2" x 3/16" x 1/8"</i></p>									
2-325F	Shunt Cable	1-1/2" x 3/16" x 1/8"	Standard P						

Equivalency  
to S.Harness  
bar  
And  
Strength of  
Attach point

# Lap & Anti-sub Structure CAD/dimensions

## Strength of Lap and Anti-sub Attach point

[illegible]

## Strength of F.Hoop mount

[illegible]

## Strength of M.Hoop mount

[illegible]

## Strength of Hoop Brace mount

## Select Structure and fill in BLANKs



# Harness Attachments

## Actual Load Test

If the Lap Belt and Anti-Submarine Belt share a fixed point, a strength of  $>30\text{kN}$  is required.  
If not shared, select "No; Separate" ( $>15\text{ kN}$  each)

### Lap Belt and Anti-Submarine Belt are mounted independently

EQ				
	Lap belt and anti-submarine attachment:	Monocoque		EQ
F.7.10.1.d	Lap and anti-sub share attachment or insert?:	No; Separate		EQ
F.7.10.1.d	Minimum spacing, lap to anti-sub 125mm (4.92in):	130	mm	EQ
F.7.10.2.a	Minimum distance, fixture to load 125mm (4.92in):	130	mm	EQ
F.7.10.1.c	Force at failure or maximum tested $\geq 15\text{kN}$ :	16000	N	EQ

EQ				
	Separate Anti-Sub:	Monocoque		EQ
	Same insert design as lap or anti-sub?:	Yes		EQ
F.7.10.1.c	Force at failure or maximum tested $\geq 15\text{kN}$ :	0	N	N/A

### Mount shared by Lap Belt and Anti-Submarine Belt

REJECT				
	Lap belt and anti-submarine attachment:	Monocoque		EQ
F.7.10.1.d	Lap and anti-sub share attachment or insert?:	Yes		EQ
F.7.10.1.d	Minimum spacing, lap to anti-sub 125mm (4.92in):	130	mm	N/A
F.7.10.2.a	Minimum distance, fixture to load 125mm (4.92in):	130	mm	EQ
F.7.10.1.d	Force at failure or maximum tested $\geq 30\text{kN}$ :	16000	N	REJECT

EQ				
	Separate Anti-Sub:			N/A
	Same insert design as lap or anti-sub?:			N/A
F.7.10.1.c	Force at failure or maximum tested $\geq 15\text{kN}$ :	0	N	N/A

N/A -> Not required

# Monocoque Attachments

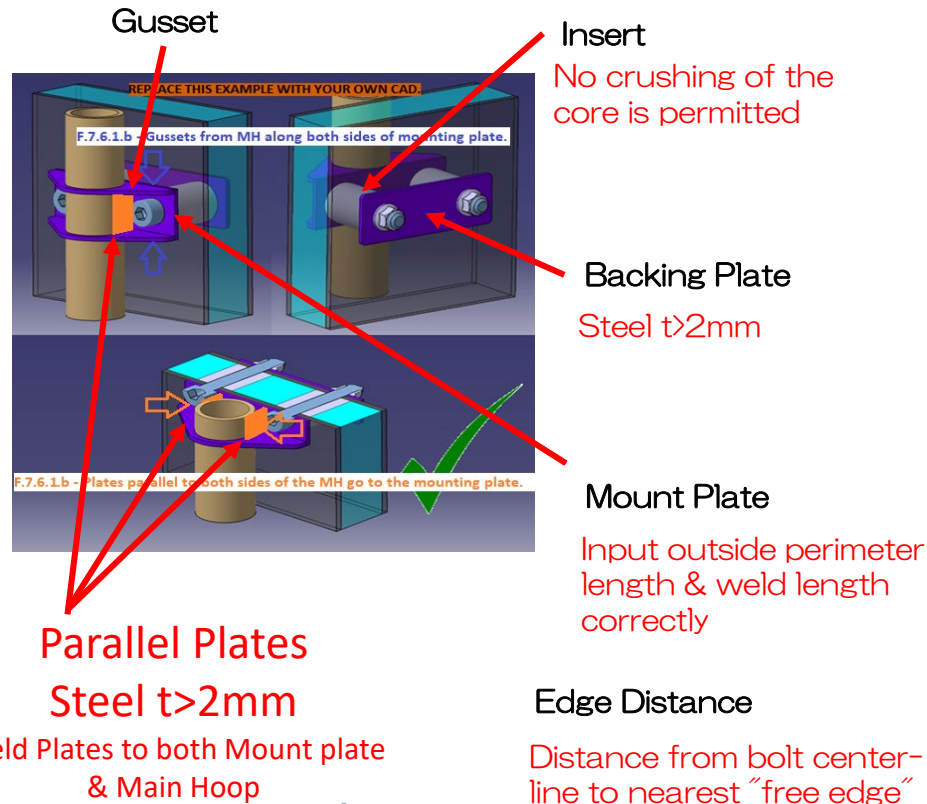
2023 revised

## Attachment point Calculation

**Attach a diagram that proves dimensions of Bracket / Insert / Backing Plate / Edge Distance**

Each attachment point requires no less than two 8 mm or 5/16" minimum diameter Critical Fasteners.

The Front Hoop may be fully laminated into the monocoque if (Partially exposed for tube thickness inspection)

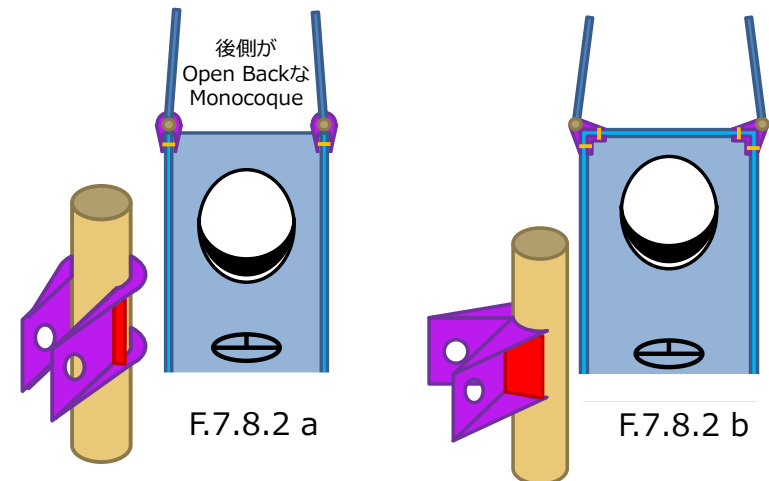


2023 added

F.7.6.1

2023 added

For Semi-Monocoque -> F.7.8.2



**Must obey one of two.**

## F.8 Front Protection

[illegible]

## Equivalency of FBH

## Select Structure and fill in BLANKs

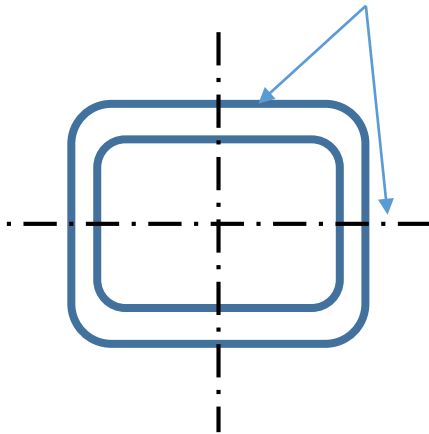
# Equivalency of Front Bulkhead

Prove equivalency to two Baseline Steel Tubes

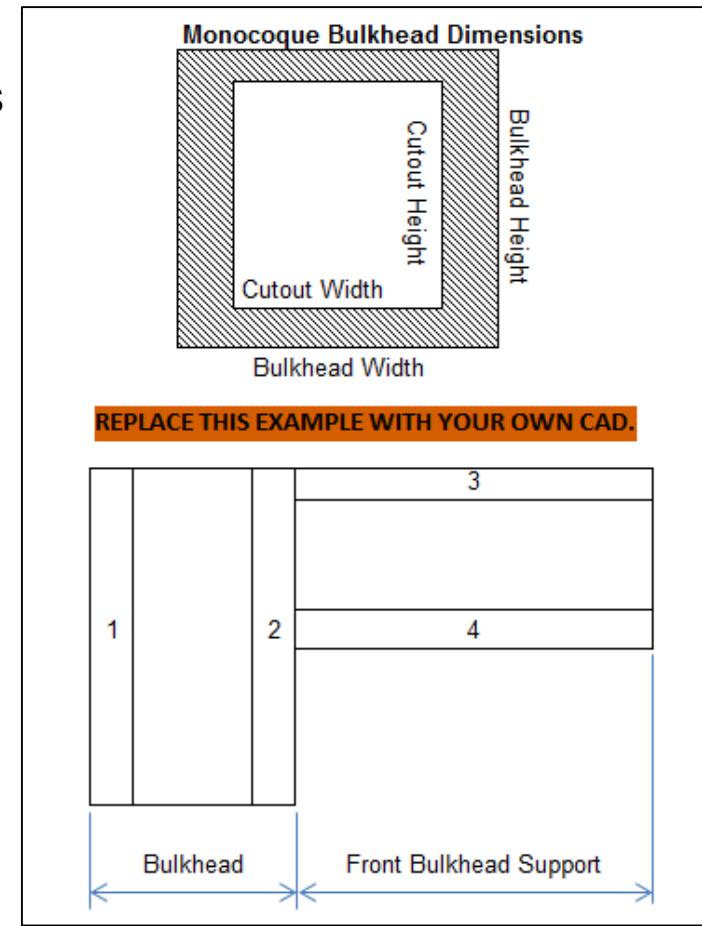
## (1) Flat panel calculation

Based on Modeled L Shape(F.7.2.1),  
it will be evaluated as equivalent 2 or 3 steel tubes

Weaker cross-section is used for calculation



Since the input value of F.7 Composite Chassis is used for the FBHS part, enter them first



F.7.2.1 L shaped Model